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Deriving a dilution of precision indicator for GNSS factor graph optimization solutions†

Paul Thevenon ¹ *, Hakim Cherfi ¹ and Julien Lesouple ¹

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- † This paper is an extended version of our paper published in ENC 2024.

Abstract: Dilution of Precision (DOP) is routinely used in GNSS to assess the quality of the constella- ¹ tion geometry for the positioning algorithm. Those DOP factors are computed from the estimation 2 covariance of a snapshot weighted least squares (WLS) estimate under certain hypotheses. This 3 paper proposes to define DOP factors for GNSS solutions based on Factor Graph Optimization (FGO). ⁴ FGO solutions have become popular in the GNSS domain. They allow to easily model probabilistic ⁵ contraints, called factors, over a large time window, by mixing observations and motion constraints ⁶ accross consecutive epochs. The solution is solved by performing a batch WLS estimation for the states at all considered epochs, using all available factors. Due to the simple nature of the estimation ⁸ algorithm – a WLS solution – it is possible to derive the theoretical estimation error covariance, which will indicate the accuracy of the computed solution. In this paper, a formula is proposed to $\frac{10}{10}$ approximate the DOP for the FGO solution. Then, the formula is validated in various scenarios $\frac{11}{10}$ involving fixed or changing satellite visibility. The same state of the state o

Keywords: GNSS; Factor Graph Optimization; Precise Positioning 13

1. Introduction 14

The positioning algorithms implemented in GNSS receivers uses estimation techniques ¹⁵ to combine GNSS observations, such as code pseudo-ranges or doppler offset measurements, to estimate the unknown position, velocity and clock bias of a GNSS receiver. Several algorithms are commonly used, such as the Weighted Least Squares estimation 18 and the Kalman Filter [1], [2]. The Weighted Least Squares (WLS) estimation applied with $\frac{1}{19}$ code pseudo-ranges is called the Single Point Positioning (SPP) solution, and is the most 20 basic solution provided by a GNSS receiver. Kalman filter-based techniques add a state ₂₁ transition model, which corresponds to a motion model for the position/velocity/clock ₂₂ states, in order to refine the estimate $[3]$. It can also be applied with phase pseudo-range 23 observations, leading the well-known Precise Point Positioning $[4]$.

Those algorithms provide both the estimate and the covariance matrix of the state 25 vector. The covariance matrix gives an indication of the uncertainty of the estimate, and its $\frac{1}{26}$ validity depends on the matching of the assumed models with the real conditions of the 27 data collect. Among the usual quality indicators, the DOP (Dilution of Precision) factors ²⁸ are often encountered [1]. The DOP factors are derived from the estimation covariance $\frac{29}{29}$ of the WLS solution, assuming that all code pseudo-ranges are affected by a Gaussian error that has the same variance and is independent from other pseudo-ranges. Given $\frac{31}{21}$ those assumptions, the DOP factors only depend on the relative satellite positions with 32 regards to the receiver. They quantify the impact of the satellite constellation geometry 33 on the uncertainty of the WLS solution. In the context of this paper, satellite constellation $\frac{34}{4}$ geometry refers both to the satellite position relative to the receiver, but also to the possible ³⁵ masking of satellites by surrounding obstacles. DOP is used for assessment of solution $\frac{36}{10}$ quality, filtering bad position estimates or planning data collections.

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More recently, Factor Graph Optimization (FGO) has been applied to the GNSS positioning problem with great success [5], [6]. FGO performs a batch WLS estimation over a $\frac{39}{2}$ large time window, as opposed to a snapshot algorithm as used in the SPP solution. In order $\frac{40}{10}$ to benefit from the optimization over a large time window, motion factors are introduced $_{41}$ in addition to GNSS observations to provide constraints between states in adjacent epochs. 42 Those motion factors are essentially similar to the state transition model in a Kalman filter. ⁴³ Therefore, all observations contribute to the accuracy of the solution at each epoch, through 44 the correlation of the estimated states across epochs introduced by the motion factors.

While the impact of the constellation geometry on the solution accuracy is well described for snapshot WLS solutions, such as the SPP solution, it is more difficult to determine how the constellation geometry will impact the solution accuracy in an FGO solution, $\frac{48}{48}$ due to the intricate interactions between the GNSS observations at each epochs and the ⁴⁹ propagation of the accuracy through the motion factors. 50

This paper aims at deriving a DOP-like factor for FGO solutions including a generic 51 motion factor, that will provide a quality assessment of a FGO solution at each epoch, based 52 on the constellation geometry and motion factor uncertainty only. $\frac{53}{2}$

2. Materials and Methods 54

2.1. DOP in a snapshot least squares solution 55

The considered positioning algorithm is the SPP algorithm, consisting in the estimation 56 of the 3D ENU position and the receiver clock bias from GNSS code observations at a single 57 epoch *k*. *j* is used to index the scalar states within the state vector: $\mathbf{x}_k = [x_{j,k}]_{j \in [\![1,4]\!]}$
The linearized observation model around an approximate position is:

The linearized observation model around an approximate position is: \sim 59

$$
\mathbf{y}_k = \mathbf{H}_k \mathbf{x}_k + \mathbf{n}_k, \quad \mathbf{n}_k \sim \mathcal{N}(\mathbf{0}, \mathbf{R}_k)
$$
(1)

where y_k is observation vector comprising all GNSS code observations at epoch k , x_k is the \qquad 60 unknown state vector, \mathbf{H}_k is the Jacobian matrix of the code observation model and \mathbf{n}_k is \quad 61 the observation error vector, assumed normally-distributed, centered with a covariance **R***^k* . ⁶²

The WLS estimate of \mathbf{x}_k [7], referred to as the *snapshot solution* is given by Eq. (2) and \mathbf{x}_s its covariance matrix is given by Eq. (3) : $\frac{64}{100}$ and $\frac{64$

$$
\hat{\mathbf{x}}_k^{(snap)} = (\mathbf{H}_k^T \mathbf{R}_k^{-1} \mathbf{H}_k)^{-1} \mathbf{H}_k^T \mathbf{R}_k^{-1} \mathbf{y}_k
$$
 (2)

$$
\mathbf{P}_k^{(snap)} = (\mathbf{H}_k^T \mathbf{R}_k^{-1} \mathbf{H}_k)^{-1}
$$
\n(3)

Assuming that the observation error is independent and identically-distributed accross 65 all available observations, i.e. $\mathbf{R}_k = \sigma_{obs}^2 \mathbf{I}$, we obtain:

$$
\mathbf{P}_k^{(snap)} = \sigma_{obs}^2 (\mathbf{H}_k^T \mathbf{H}_k)^{-1}
$$
 (4)

where σ_{obs} is the standard deviation of the GNSS code observations. σ_{obs}

This expression separates the contribution of the measurement noise from the geom- 68 etry of the constellation. We note $D_k = (\mathbf{H_k}^T \mathbf{H_k})^{-1}$ the DOP matrix. The different DOP 69 factors are computed from $\mathbf{D}_k.$ For example, assuming that the 3D position is placed on the \quad π first 3 elements of the state vector \mathbf{x}_k : \vdots 71

 Global DOP: $GDOP = \text{Trace}(\mathbf{D}_k)$) and the contract of $\overline{}$ $\overline{\phantom$

• Position DOP:
$$
PDOP = (D_k)_{1,1} + (D_k)_{2,2} + (D_k)_{3,3}
$$

where $(A)_{i,j}$ refers to the element at row *i* and column *j* of a matrix **A**.

2.2. Factor Graph model 75

Figure 1. Factor graph considered in this study.

In this study, a simple factor graph (Figure 1) is considered for one *scalar* state x_j at τ ⁶ different epochs indexed by $k \in [1, K]$. $x_{j,k}$ refers to one of the scalar state at epoch k , such τ as the position along one axis or the receiver clock bias. $\frac{78}{18}$

We assume that the snapshot LS algorithm provides a solution for each of the K_{η} epochs of a trajectory. From each snapshot solution, the snapshot estimation accuracy 80 $\sigma_{j,k}^2 = (\mathbf{P}_k^{(snap)})$ $\binom{sin(\mu)}{k}$ *j*,*j* is used as a factor to constrain the scalar state x_j at epoch *k* and is referred 81 to as *snapshot GNSS factor*.

$$
y_{j,k} = x_{j,k} + n_{j,k}, \quad n_{j,k} \sim \mathcal{N}(0, \sigma_{j,k}^2)
$$
 (5)

An additional factor is introduced to model the probabilistic constraint linking the $\frac{1}{83}$ scalar states at two consecutive epochs. It assumes that the evolution of the scalar state ⁸⁴ between 2 epochs is known, with an uncertainty assumed to be normally-distributed with $\frac{1}{100}$ a standard deviation *q*. This constraint is referred to as a *motion factor* in this document. ⁸⁶ This generic motion factor uses the following affine model: 87

$$
m_{k,k+1} = x_{j,k+1} - (x_{j,k} + b_{k+1}) + n_{m_{k,k+1}}, \quad n_{m_{k,k+1}} \sim \mathcal{N}(0, q^2)
$$
 (6)

where $n_{m_{k,k+1}}$ accounts for the uncertainty of the motion constraint, assumed to be a centered normal random variable with a constant standard deviation q and b_{k+1} is a known \square s parameter, e.g. an integrated velocity coming from another sensor. **900** 90

Examples of motion constraints are the estimation of the receiver displacement by $_{91}$ GNSS Doppler or Time Difference of Carrier Phase measurements [6], estimation of the $\frac{92}{2}$ receiver displacement by an external sensor such as an inertial measurement unit or a 93 visual simultaneous localization and mapping solution [2], or simply a dynamic model, $_{94}$ such as a random walk or constant velocity model with Gaussian uncertainty [3]. $\qquad \qquad \text{ss}$

Finally, it is assumed that the errors of the snapshot GNSS factors and those of the 96 motion factors are independent between each other and across epochs. Therefore, when $\frac{97}{97}$ stacking all the factors in a single vector **y**, we obtain the following estimation problem: ⁹⁸

$$
\mathbf{y} = \begin{bmatrix} y_{j,1} \\ y_{j,2} \\ \vdots \\ y_{j,K} \\ m_{1,2} \\ m_{2,3} \\ \vdots \\ m_{K-1,K} \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & \dots & 0 \\ 0 & 1 & 0 & \dots & 0 \\ 0 & 0 & 1 & \ddots & \vdots \\ \vdots & \vdots & \ddots & \ddots & 0 \\ 0 & 0 & \dots & 0 & 1 \\ -1 & 1 & 0 & \dots & 0 \\ 0 & -1 & 1 & \ddots & \vdots \\ \vdots & \ddots & \ddots & \ddots & 0 \\ 0 & \dots & 0 & -1 & 1 \end{bmatrix} \begin{bmatrix} x_{j,1} \\ x_{j,2} \\ \vdots \\ x_{j,K} \end{bmatrix} + \begin{bmatrix} n_{j,1} \\ n_{j,2} \\ \vdots \\ n_{j,K} \\ n_{m_{1,2}} \\ \vdots \\ n_{m_{N-2}} \\ n_{m_{N-3}} \end{bmatrix}
$$

$$
= \underbrace{\begin{bmatrix} \mathbf{I}_{K} \\ \mathbf{K} \\ \mathbf{H}_{m} \end{bmatrix}}_{\mathbf{H}} \mathbf{x}_{j,1:K} + \underbrace{\begin{bmatrix} \mathbf{n}_{j,1:K} \\ \mathbf{n}_{j,1:K} \\ \mathbf{n}_{m_{1:K}} \end{bmatrix}}_{\mathbf{n}_{j}}, \quad \mathbf{n}_{j} \sim \mathcal{N} \left(\mathbf{0}_{2K-1}, \begin{bmatrix} \mathbf{diag}(\sigma_{j,1}^{2}, \dots, \sigma_{j,K}^{2}) & \mathbf{0}_{K \times K-1} \\ \mathbf{0}_{K-1 \times K} & \sigma_{j}^{2} \mathbf{I}_{K-1} \end{bmatrix} \right) \quad (7)
$$

2.3. Batch least squares solution of the considered Factor Graph ⁹⁹

When ignoring the motion constraints in Eq. (7) , we find the trivial solution which is $_{100}$ called the *snapshot* **solution**. For this particular solution, the snapshot covariance matrix ¹⁰¹ for state x_j accross all epochs from 1 to *K* is: 102

$$
\mathbf{P}_{j}^{(snap)} = \mathbf{diag}(\sigma_{j,1}^{2}, \dots, \sigma_{j,K}^{2})
$$
\n(8)

The solution of the factor graph mixing both snapshot GNSS factors and motion factors 103 is referred to as the *batch* **solution**. The batch solution covariance matrix is obtained with $_{104}$ the following equation. The set of the following equation.

$$
\mathbf{P}_{j}^{(batch)} = (\mathbf{H}^{T} \mathbf{Cov}(\mathbf{n}_{j})^{-1} \mathbf{H})^{-1} = \left([\mathbf{I}_{K} \quad \mathbf{H}_{\mathbf{m}}^{T}] \left[\frac{\mathbf{P}_{j}^{(snap)} \mid \mathbf{0}_{K \times K - 1}}{\mathbf{0}_{K - 1 \times K} \mid q^{2} \mathbf{I}_{K - 1}} \right]^{-1} \left[\mathbf{I}_{K} \right] \right)^{-1}
$$

$$
= \left((\mathbf{P}_{j}^{(snap)})^{-1} + \frac{1}{q^{2}} \mathbf{H}_{\mathbf{m}}^{T} \mathbf{H}_{\mathbf{m}} \right)^{-1}
$$
(9)

The inverse of the batch solution covariance is a so-called a tri-diagonal matrix $[8]$. $\frac{106}{2}$

$$
\mathbf{P}_{j}^{(batch)} = \begin{bmatrix} \frac{1}{q^{2}} + \frac{1}{\sigma_{j,1}^{2}} & -\frac{1}{q^{2}} & 0 & \dots & 0\\ -\frac{1}{q^{2}} & \frac{2}{q^{2}} + \frac{1}{\sigma_{j,2}^{2}} & -\frac{1}{q^{2}} & \ddots & \vdots\\ 0 & \ddots & \ddots & \ddots & 0\\ 0 & \ddots & -\frac{1}{q^{2}} & \frac{2}{q^{2}} + \frac{1}{\sigma_{j,k-1}^{2}} & -\frac{1}{q^{2}}\\ 0 & \dots & 0 & -\frac{1}{q^{2}} & \frac{1}{q^{2}} + \frac{1}{\sigma_{j,k}^{2}} \end{bmatrix}^{-1}
$$
(10)

2.4. Efficient computation of the batch estimation covariance

To compute efficiently the batch solution covariance matrix, one can use the particular $_{108}$ shape of (10). Additionally, we are interested only in the diagonal elements of the matrix, $_{109}$ to characterize the uncertainty of the considered scalar state estimate at each epoch *k*. The ¹¹⁰ symbolic expression of the diagonal elements for the first few values of K is shown in $\frac{111}{111}$ Annex A. It is observed that the *k*-th diagonal element is a ratio of 2 expressions:

$$
(\mathbf{P}_j^{(batch)})_{k,k} = \frac{N_{j,k}(q, \sigma_{j,1}, \dots, \sigma_{j,K})}{D_j(q, \sigma_{j,1}, \dots, \sigma_{j,K})}
$$
(11)

where $N_{j,k}$ and D_j are expressions depending on $q, \sigma_{j,1}, \ldots, \sigma_{j,K}$ 113

The denominator D_j is common to all epochs of the considered trajectory. It can be 114 computed using a second-order linear recurrence relation: ¹¹⁵

$$
p_0 = \frac{1}{q^2}, \quad p_1 = 1 + \frac{\sigma_{j,1}^2}{q^2}
$$

\n
$$
p_k = (q^2 + 2\sigma_{j,k}^2)p_{k-1} - \sigma_{j,k}^2 \sigma_{j,k-1}^2 p_{k-2} \quad \text{, for } 2 \le k < K
$$

\n
$$
D_j = p_K = (q^2 + \sigma_{j,K}^2)p_{K-1} - \sigma_{j,K}^2 \sigma_{j,K-1}^2 p_{K-2} \tag{12}
$$

The numerator $N_{j,k}$ depends on the considered epochs within the trajectory and can $_{116}$ be computed using the following expression: 117

$$
N_{j,k} = D_j(\sigma_{j,k} = 0)\sigma_{j,k}^2
$$
\n(13)

where $D_j(\sigma_{jk} = 0)$ is the evaluation of expression D_j at $\sigma_{jk} = 0$ and leaving other parameters untouched. The state of the state o

The expressions of $N_{i,k}$ and D_i have been validated by comparing the diagonal elements of Eq. (10) to the values computed using Eq. (11), using a symbolic mathematical $_{121}$ software up to $K = 10$, and numerical simulations up to $K = 100$. Elements of the 122 mathematical demonstration to obtain those expressions are available in Annex B.

2.5. DOP computation for a Factor Graph Optimization solution

By definition, the DOP factor is defined as the ratio between the uncertainty of the 125 combined estimated states and the uncertainty of the GNSS observations, assuming that all ₁₂₆ GNSS observations are independent and affected by an error with the same variance σ_{obs}^2 : 127

$$
DOP_k = \frac{\sqrt{\sum_{j \in S} \text{var}(\hat{x}_{j,k})}}{\sigma_{obs}}
$$
(14)

where $var(\hat{x}_{j,k})$ is the estimation variance of state $x_{j,k}$, and S is the set of indexes of the 128 scalar states of interest. For example, if the state vector at a particular epoch is composed 129 of [*east*, *north*, *up*, *clock*_*bias*], to compute the horizontal DOP, one has to combine the east ¹³⁰ and north position uncertainty and $S = [1, 2]$.
The previous section provided the estimation uncertainty computation for both snap-

The previous section provided the estimation uncertainty computation for both snapshot and batch solutions, for a scalar element of the state vector. The computation can then $_{133}$ be performed for the scalar states of interest before combining them into a DOP factor.

We define the DOP factors coming from the snapshot solution, noted $DOP_k^{(snap)}$, and 135 the one coming from the batch solution including a motion factor, noted $DOP_k^{(batch)}$

$$
DOP_k^{(snap)} = \frac{\sqrt{\sum_{j \in S} (\mathbf{P}_k^{(snap)})_{j,j}}}{\sigma_{obs}} = \sqrt{\sum_{j \in S} (\mathbf{H}_k^T \mathbf{H}_k)_{j,j}}
$$
(15)

$$
DOP_k^{(batch)} = \frac{\sqrt{\sum_{j \in S} (\mathbf{P}_k^{(batch)})_{j,j}}}{\sigma_{obs}} \approx \frac{\sqrt{\sum_{j \in S} \frac{N_{j,k}(q,\sigma_1,\ldots,\sigma_K)}{D_j(q,\sigma_1,\ldots,\sigma_K)}}}{\sigma_{obs}}
$$
(16)

Note that while the snapshot DOP does not depend on σ_{obs} , the batch DOP depends $_{137}$ on the ratio between the motion uncertainty and the observation uncertainty q/σ_{obs} .

3. Results 139

3.1. Validation method of the proposed batch DOP formula ¹⁴⁰

The formula, Eq. (11) , to obtain uncertainty for a batch solution from the snapshot $_{141}$ scalar uncertainty has been validated for a single scalar state.

However, when combining the uncertainty of several scalar states to compute a DOP $_{143}$ factor, the correlated impact of an observation error on the different states should be 144 accounted. This is not the case for the batch DOP computed using Eq. (16) , since the 145 diagonal elements of the covariance matrix of the batch solution are approximated by ¹⁴⁶ considering each scalar state uncertainty independently from the other states. ¹⁴⁷

The following subsections aim at quantifying the impact of this approximation. To do $_{148}$ so, the batch DOP obtained from Eq. (16) and the exact batch DOP, obtained by computing $_{149}$ the covariance of the full state vectors with Eq. (10), are compared based on the difference 150 of HDOP and the ratio of HDOP. The results are summarized in Table 1. ¹⁵¹

Table 1. HDOP difference and ratio between proposed formula and exact computation.

Scenario	Factor graph length K	Motion to observation uncertainty ratio q/σ	Max HDOP difference	Max HDOP ratio
fixed geometry	[10, 50]	[0.01, 0.1, 0.5, 1]	0.008	$\%$
varying geometry	50	[0.01, 0.1, 0.5, 1]	0.02	3%
urban scenario	1009	[0.01, 0.1, 0.5]	0.15	22%

When considering several consecutive epochs at a high rate (typically higher than 0.1_{153} Hz), the constellation geometry can be considered as fixed over a window of a few tens 154 of epochs. In the case of an open sky receiver, the snapshot DOP will typically remain ¹⁵⁵ constant, except when one satellite appear or disappear in the antenna's field of view.

Figure 2. HDOP comparison for fixed **(a)**,**(b)** and time-varying **(c)** constellation geometry

Figure 2 (a), (b) shows the batch HDOP for all epochs for a mix of factor graph length $_{157}$ *K* and motion to observation uncertainty ratio q/σ . The HDOP value computed with the 158 formula is close to the exact HDOP value, especially for low values of q/σ .

On the general shape of the batch DOP, we can observe that the DOP is lower in the 160 middle of the graph. This is due to the fact that all surrounding epochs contribute to the $_{161}$ accuracy at a particular epoch. Secondly, the overall accuracy depends on the value of q/σ : 162 the lower q/σ is, the better the accuracy is. Finally, for larger q/σ , the DOP usually reaches 163 a floor value for the central part of the trajectory that is common for different values of *K*. ¹⁶⁴ This points towards the fact that increasing the factor graph length may not result in better ¹⁶⁵ accuracy once the floor has been reached. 166

3.3. Validation of the FGO DOP formula in a varying constellation case 167

In this section, the nominal GPS constellation from IS-GPS-200M is considered. The 168 scenario considers a fixed receiver in Toulouse on the day 2023-01-01 with a 5-min interval over the first hours of the day. This scenario allows to have varying and realistic $\frac{170}{170}$ constellation geometries, leading to temporal variations of the snapshot DOP. ¹⁷¹

Figure 2 (c) shows the exact HDOP and the HDOP computed using equation (16) for a ¹⁷² fixed value of $K = 50$ and various values of q/σ . Again, the HDOP values obtained with 173 the formula are close to the exact values. 174

An interesting observation can be made. For high accuracy motion constraints (e.g. 175 $q/\sigma \leq 0.1$), variations of HDOP are attenuated and the batch HDOP reaches a floor value 176 for central epochs. In this particular cases, the accurate motion constraints allow to have a 177 good relative accuracy between consecutive position estimates, and the observations from 178 every epoch to contribute to the absolute accuracy of the whole trajectory. ¹⁷⁹

3.4. Validation of the FGO DOP formula in constrained urban case 180

In this section, we study the case where the constellation geometry is changing in time $_{181}$ due to the masking by building as a GNSS receiver is moving across a city. The considered 182 trajectory comes from the Google Smartphone Decimeter Challenge 2021 [9], and has 183 been chosen as one that passes through an urban center. Figure 3 shows the 2D trajectory. 184 Frequent satellite masking occur leading to frequent snapshot HDOP degradation.

Figure 4 shows the exact HDOP and the HDOP computed using Eq. (16) for $K = 1009$, 186 i.e. all available epochs at 0.5 Hz, and various values of q/σ . Again, the HDOP values $_{187}$ obtained with the formula are close to the exact values. The largest difference corresponds ¹⁸⁸ to epochs where fewer satellites are available. At those epochs, sparse GNSS observations 189 results in larger correlation between the east and north position estimation errors, and 190 therefore, larger approximation error of the proposed formula. 191

Figure 3. Trajectory "2021-04-28-US-SJC-1/Pixel4" from the GSDC2021 dataset.

Figure 4. HDOP comparison for urban masking scenario

The FGO solution improves the accuracy of the solution, even in the epochs without $_{192}$ enough satellites to compute a solution. For those epochs, the snapshot uncertainty has 193 been clipped to a value of 100 m for each position state. Again, the temporal DOP variations $_{194}$ are attenuated in the case of highly accurate motion uncertainties. 195

4. Discussion 196

This paper proposes a definition of the notion of DOP for batch solution including 197 a generic motion factor, applicable to FGO solutions. The obtained DOP factor mainly 198 depends on the snapshot DOP, the FG length *K* and the motion to observation uncertainty ¹⁹⁹ ratio q/σ . Depending on q/σ , the batch DOP reaches a floor value for large FG length, $_{200}$ meaning that increasing the FG length provides low marginal gains in terms of accuracy. ₂₀₁

A formula is derived for fast computation of batch DOP factors. The approximation 202 error comes from ignoring the correlated impact of observation errors on multiple states $_{203}$ when computing the scalar uncertainty for each state. The error on HDOP is of the order of $_{204}$ a few percents in open sky conditions, and rises to 22% in constrained conditions, when a ₂₀₅ lower number of satellites is available. 206

Based on this formula, a future work could try to derive an optimal factor graph length $_{207}$ depending on the satellite visibility conditions of consecutive epochs, in order to limit the $_{208}$ computational complexity of large factor graph solution. ²⁰⁹

The formula could also be improved by considering partial observability of the state 210 vector, where the snapshot solution cannot be computed (e.g. when less than 4 satellites $_{211}$ are available). Solutions to consider the covariance between the scalar state of the snapshot ₂₁₂ solution could also be investigated in order to reduce the approximation error.

Finally, the proposed formula could be applied to other types of states, such as the 214 float carrier ambiguities. This could provide insight to optimize factor graph length of a 215 float solution, before trying to fix the carrier ambiguities. 216

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Appendix A. Symbolic expression of the first few diagonal elements of $\mathbf{P}^{(batch)}_j$

In the annexes, a simplified notation will be used by dropping the index *j* referring to 225 a particular scalar state in the state vector: $P^{(batch)} \triangleq P^{(batch)}_i$ j ^(batch) and $\sigma_k \triangleq \sigma_{j,k}$ \cdot 226

It can be observed that the *k*-th diagonal element of $P^{(batch)}$ is a ratio of 2 expressions: 227

$$
(\mathbf{P}^{(batch)})_{k,k} = \frac{N_k(q, \sigma_1, \dots, \sigma_K)}{D(q, \sigma_1, \dots, \sigma_K)}
$$

where N_k and *D* are expressions depending on $q, \sigma_1, \ldots, \sigma_K$ 228 The first few diagonal elements of *P* (*batch*) are given in Table A1. ²²⁹

Table A1. Symbolic expression of the diagonal elements of $P_i^{(batch)}$ $\int_{i}^{(outch)}$ for the first values of *K*.

Appendix B. Computation of the denominator of the elements of $\mathbf{P}^{(batch)}_j$

The denominator D of the elements of $P^{(batch)}$ is equal to the determinant of $(P^{(batch)})^{-1}$. ²³¹ This comes directly from the Laplace formula to compute the inverse of a matrix.

This section reveals how to compute the determinant of matrix $(\mathbf{P}^{(batch)})^{-1}$. The partic- 233 ular shape of ($\mathbf{P}^{(batch)}$)⁻¹ allows to have interesting expressions to compute its determinant. 234 Indeed, $(\mathbf{P}^{(batch)})^{-1}$ is a tri-diagonal matrix, i.e. a banded matrix with 3 non-zero diagonals 235 elements. For such matrix, a recurrence formula exist to compute its determinant. 236

Theorem A1 (Determinant of a tri-diagonal matrix)**.** *(Muir, 1960) [8]* ²³⁷

Let us consider the following $K \times K$ *tri-diagonal matrix*

 a_1 b_1 0 ... *a*₁ *b*₁ 0 ... 0 0 c_1 a_2 b_2 ... 0 0 *c*² *a*³ . . . 0 0 *.* 0 0 0 ... a_{K-1} b_{K-1} 0 0 0 . . . *cK*−¹ *a^K* ı

The determinant of such matrix, noted dK, can be computed with the following recurrence relation: ²³⁹

$$
d_0 = 1, \quad d_1 = a_1
$$

$$
d_k = a_k d_{k-1} - b_{k-1} c_{k-1} d_{k-2} \quad , \text{ for } 2 \le k \le K
$$

When applying this theorem to $(\mathbf{P}^{(batch)})^{-1}$, we obtain a ratio of 2 expressions where 240 the denominator is always $q^{2(K-1)} \prod_{n=1}^{K} \sigma_n^2$. Therefore, noting the numerator p_K , we have: 241

$$
d_K = \frac{p_K}{q^{2(K-1)} \prod_{n=1}^K \sigma_n^2}
$$

As we may have to evaluate this expression for $\sigma_n = 0$, it is more convenient to work σ_{n+1} with $p_K = d_K q^{2(K-1)} \prod_{n=1}^K \sigma_n^2$.

Lemma A1 (recurrence relation considering the sequence $s_k = d_k \prod_{j=1}^k (q^2 \sigma_j^2)$).

$$
s_0 = 1, \quad s_1 = q^2 + \sigma_1^2
$$

\n
$$
s_k = (q^2 + 2\sigma_k^2)s_{k-1} - \sigma_k^2\sigma_{k-1}^2s_{k-2} \quad , \text{for } 2 \le k < K
$$

\n
$$
s_K = (q^2 + \sigma_K^2)s_{K-1} - \sigma_K^2\sigma_{K-1}^2s_{K-2}
$$

Proof. 244

$$
s_k = d_k \prod_{j=1}^k (q^2 \sigma_j^2)
$$

= $(a_k d_{k-1} - b_{k-1} c_{k-1} d_{k-2}) \prod_{j=1}^k (q^2 \sigma_j^2)$ (using Theorem A1)
= $a_k q^2 \sigma_k^2 d_{k-1} \prod_{j=1}^{k-1} (q^2 \sigma_j^2) -b_{k-1} c_{k-1} q^4 \sigma_k^2 \sigma_{k-1}^2 d_{k-2} \prod_{j=1}^{k-2} (q^2 \sigma_j^2)$
= $a_k q^2 \sigma_k^2 s_{k-1} - \sigma_k^2 \sigma_{k-1}^2 s_{k-2}$ (replacing b_k and c_k by their values)

Theorem A1 provides the initial values of the sequence d_k , which are used to define $\frac{245}{2}$ the initial values of s_k . . ²⁴⁶

The term p_K is actually equal to $p_K = \frac{1}{q^2} d_K \prod_{j=1}^K (q^2 \sigma_j^2) = \frac{1}{q^2} s_K$. The common factor 247 to all terms $\frac{1}{q^2}$ is applied to the initial values of a new sequence p_k . Finally, we can also ₂₄₈ replace by the values of a_k . Then, we have: 249

$$
p_0 = \frac{1}{q^2} d_0 = \frac{1}{q^2}, \quad p_1 = \frac{1}{q^2} d_1 q^2 \sigma_1^2 = 1 + \frac{\sigma_1^2}{q^2}
$$

\n
$$
p_k = (q^2 + 2\sigma_k^2) p_{k-1} - \sigma_k^2 \sigma_{k-1}^2 p_{k-2} \quad \text{for } 2 \le k < K
$$

\n
$$
p_K = (q^2 + \sigma_K^2) p_{K-1} - \sigma_K^2 \sigma_{K-1}^2 p_{K-2}
$$

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